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Research article

Научная статья



## Structure and properties of welds in electron beam welding of iron-chromium-nickel alloy EP718

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**Abstract:** This article presents the results of a study focused on the formation of structural characteristics and properties of welded joints in the EP718 alloy with a 13 mm thickness (accounting for a 3 mm technological substrate). The study explores variations in electron beam welding parameters, such as beam current and the speed of its movement across the specimen's surface, to determine the optimal welding mode for this alloy. This alloy is crucial in the production of high-pressure stators for aircraft engines, as the component operates under low-cycle loads at high stress levels, making its performance critical. Specimens that were welded with a beam speed ( $v$ ) of 0.0042 m/s and a beam current ( $i$ ) of 85 mA exhibited a minimum tensile strength of 1160 MPa. On the other hand, specimens welded with  $v = 0.006$  m/s and  $i = 65$  mA demonstrated a maximum tensile strength of 1270 MPa. However, it's noteworthy that specimens welded at 0.006 m/s with beam currents of 120 mA and 75 mA experienced fracture along the weld, while specimens welded at 0.006 m/s with a beam current of 65 mA and at 0.0042 m/s with a beam current of 85 mA exhibited fracture in the heat-affected zone at a distance of 0.5–3.0 mm from the weld. Examination of the structure of specimens welded at  $v = 0.006$  and 0.0042 m/s and  $i = 120$  mA, 75 mA, and 85 mA revealed expanded grain boundaries in the heat-affected zone. Consequently, the optimal welding mode was identified as having a beam speed of 0.006 m/s and a beam current of 65 mA. In this mode, no thickened grain boundaries were detected, and a maximum tensile strength of 1270 MPa was achieved.

**Keywords:** electron beam welding, heat resistant nickel alloys, heat affected zone, structure, properties.

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## Структура и свойства сварных швов при электронно-лучевой сварке железохромоникелевого сплава ЭП718

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**Аннотация:** Приведены результаты исследований особенностей формирования структуры и свойств сварных соединений сплава ЭП718 толщиной 13 мм (с учетом технологической подкладки 3 мм) за счет варьирования параметров электронно-лучевой свар-

ки (тока луча и скорости его перемещения по поверхности образца) и определения оптимального режима сварки для данного сплава, используемого при изготовлении статора высокого давления авиационного двигателя. Деталь является ответственным крупногабаритным изделием сложной профильной формы и работает в условиях малоциклических нагрузок при высоком уровне напряжений. Минимальный предел прочности 1160 МПа имеют образцы, сваренные при скорости перемещения луча по поверхности образца  $v = 0,0042$  м/с и токе луча 85 мА. Для образцов, сваренных при  $v = 0,006$  м/с и  $i = 65$  мА, характерен максимальный предел прочности, равный 1270 МПа. При определении временного сопротивления у образцов, сваренных при  $v = 0,006$  м/с,  $i = 120$  и 75 мА, разрушение произошло по сварному шву, а у образцов, сваренных при  $v = 0,006$  м/с,  $i = 65$  мА и  $v = 0,0042$  м/с,  $i = 85$  мА, — по зоне термического влияния на расстоянии 0,5–3,0 мм от сварного шва. При микроисследовании структуры образцов, сваренных при  $v = 0,006$  и 0,0042 м/с и  $i = 120$ , 75 и 85 мА соответственно, выявлены расширенные границы зерен в зоне термического влияния. Таким образом, оптимальным является режим сварки при скорости перемещения луча по поверхности образца 0,006 м/с и токе луча 65 мА. На данном режиме утолщенных границ зерен не обнаружено и достигается максимальный предел прочности 1270 МПа.

**Ключевые слова:** электронно-лучевая сварка, жаропрочные никелевые сплавы, зона термического влияния, структура, свойства.

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## Introduction

In recent years, the use of heat-resistant alloys in heavy-duty, large-scale power units and assemblies within aviation engineering has significantly expanded. These components often operate under low cycle loads at high stress levels. Among the most compelling alloys are those with tensile strengths ranging from 980 to 1200 MPa [1]. One such alloy, KhN45VMYTUBR-ID (EP718), has gained attention due to its application in parts operating in temperature ranges of up to 700–800 °C. Welding plays a pivotal role in the production of these components.

Welded joints possess distinct mechanical and operational characteristics, which can markedly differ from those of the base metal. The reliability and longevity of welded structures are predominantly influenced by the quality and structure of the weld metal, the heat-affected zone, and the joint's design. Welding heat-resistant alloys introduces unique challenges, including the propensity for defect formation, such as cracks [2; 3].

Electron beam welding (EBW) has emerged as a method offering minimal remelted material, minimal product deformation, and strength on par with the base material. However, achieving high-quality welded joints through EBW requires precise control of edge penetration, as well as minimizing chemical and structural inhomogeneities in the weld metal and heat affected zone (HAZ) while reducing residual stresses and eliminating defects like cracks and pores [4; 5].

Analysis of scientific publications from both foreign and domestic scientists reveals that various methods for creating defect-free permanent joints in the heat-resistant alloy EP718 are actively under investigation world-

wide. For example, in [6], defect-free joints on 5 mm thick sheet specimens were achieved using CO<sub>2</sub> laser welding. In another study, laser welding (LW) of 1.3 mm thick EP718 alloy revealed defects in the root part of the weld, such as pores [7]. [8] explored welds of 3.4 mm thick specimens made by additive growth through electric arc welding (EAW) using a non-melting tungsten electrode in a shielding gas environment. Although a fine-grained structure was obtained after heat treatment, cracks were found in the HAZ. In [9; 10], the influence of heat input during electron beam welding on the possibility of eliminating microcracks in the weld and the HAZ of 2 mm thick specimens was investigated. Furthermore, [11–13] examined welds of Inconel 718 heat-resistant alloy produced through EBW, EAW, and LW technologies at various welding modes, showcasing the impact of grain size on mechanical properties. The effects of stirring during friction welding on the mechanical properties of EP718 alloy welds were explored in [14], as were the electric arc modes for welding 2 mm thick specimens [15]. Additionally, [16] delved into the intricacies of weld formation on 3 mm thick specimens using EBW, while [17] analyzed the welding of consumable electrodes by a magnetically compressed arc on 2 mm thick sheet material of the EP718 alloy.

Some publications have focused on investigating the unique aspects related to the geometry and microstructure of welds, as well as the formation of defects in both the weld and the heat-affected zone, examining ultra-high frequency and conventional EBW, as well as microwave welding [18–20].

In the studies conducted by Russian researchers [21; 22], the structure and properties of welds created through

various welding methods, such as robotic welding with a consumable electrode and LW, were thoroughly examined.

For the fabrication of permanent joints using the heat-resistant alloy KhN45MVTYUBR-ID (EP718), concentrated energy flows, including electron beam welding, show promise due to their ability to produce narrow welds and high-quality joints [23]. Nevertheless, the EBW process comes with its set of challenges and unresolved issues, including the potential for cracks, pores, and other defects within the welded metal, which can compromise the reliability and performance characteristics of the final product [24].

The primary objective of this study was to investigate the specific factors influencing the structure and properties of welded joints crafted from the EP718 alloy, featuring a thickness of 13 mm (accounting for a technological substrate of 3 mm). This was achieved by varying the parameters of electron beam welding, such as beam current and the speed of its movement across the specimen's surface, to identify the optimal welding mode. This alloy is extensively used in the production of high-pressure stators for aircraft engines, where the components are of significant size, possess intricate profiles, and operate under low cycle loads at high stress levels. Remarkably, this study considered these critical operational conditions, which were not always accounted for in previous research [6–22].

## Materials and methods

The high-pressure stator is fabricated through the welding of two components: the rear flange, which is constructed from a solid rolled ring following Industry standard OST 1.90396-91, and the middle ring, crafted from EP718 sheet material in accordance with Specifications TU 14-1-5095-92. Following the welding process, the component undergoes heat treatment to relieve stress and attain the necessary mechanical properties.

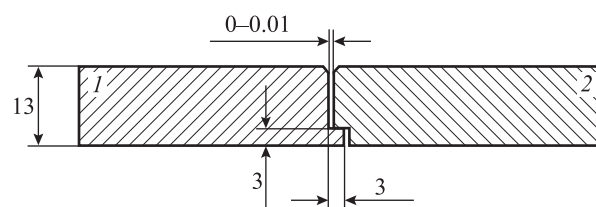
The investigation of welding parameters and the examination of microstructure and mechanical properties were conducted using simulator specimens measured 200 mm in length, 50 mm in width, and 13 mm in thickness. They are constructed from a solid rolled ring (Fig. 1, loc. 1) and EP718 heat-resistant alloy sheet material (Fig. 1, loc. 2).

The local chemical composition of the alloy was assessed through qualitative and quantitative X-ray spectral microanalysis using a scanning electron microscope. Spectral analysis was conducted to determine the chemical composition of the specimens, and the results are provided in Table 1. The material grade of the specimens matches the EP718 alloy.

Permanent connections were created using an electron beam welding facility. Various EBW modes were employed, involving adjustments to the specimen's movement speed relative to the beam and the beam current strength.

Following the welding process, the specimens underwent heat treatment. Hardening was conducted in an elevator, and ageing was carried out in a shaft electric furnace. The specific heat treatment modes, including hardening and ageing, are detailed Table 2.

To assess the mechanical properties of welded joints, specimens were prepared based on simulator specimens



**Fig. 1.** Sketch of a simulator specimen

1 – solid rolled ring; 2 – sheet material

Dimensions are given in mm

**Рис. 1.** Эскиз образца-имитатора

1 – цельнокатаное кольцо, 2 – листовой материал

Размеры указаны в мм

**Table 1. Chemical composition (wt.%) of simulator specimens' material**

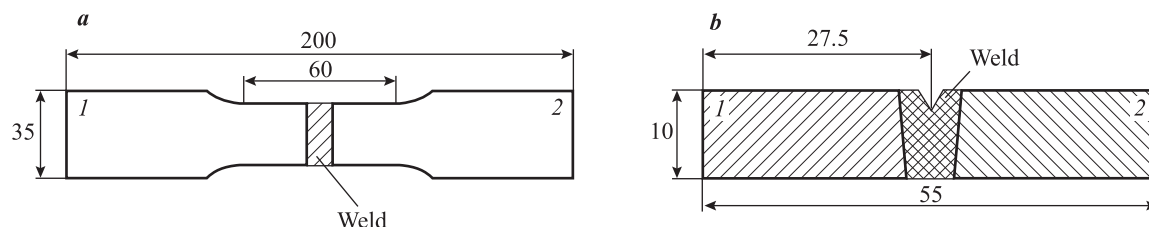
**Таблица 1. Химический состав (мас.%) материала образцов-имитаторов**

Cutting location (Fig. 1)	C	Si	Mn	S	P	Cr	Ni	W	Mo	Ti	Al	Nb
Location 1	0.055	0.007	0.07	0.0016	0.004	15.6	45.24	3.44	4.45	2.08	0.99	1.13
Location 2	0.058	0.08	0.24	0.0016	0.01	15.9	44.73	3.23	4.11	2.17	0.98	1.01
Specifications TU-14-1-5095-92	≤0.1	≤0.3	≤0.6	≤0.01	≤0.015	14–16	43–47	2.5–3.5	4.0–5.2	1.9–2.4	0.9–1.4	0.8–1.5

Table 2. Heat treatment modes for specimens before and after welding

Таблица 2. Режимы термической обработки образцов до и после сварки

Heat treatment before welding	Heat treatment after welding		
Quenching	Quenching (stress relieving)	Ageing 1	Ageing 2
$t = 1100\text{ }^{\circ}\text{C}$ , $\tau = 2\text{ h}$ , cooling in air	Heating in air $t = 1100\text{ }^{\circ}\text{C}$ , $\tau = 1\text{ h}$ , cooling to $500\text{--}600\text{ }^{\circ}\text{C}$ in a container in air with argon supply, further – without argon supply	$t = 780\text{ }^{\circ}\text{C}$ , $\tau = 5\text{ h}$ , cooling in air	$t = 650\text{ }^{\circ}\text{C}$ , $\tau = 16\text{ h}$ , cooling in air

**Fig. 2.** Sketches of specimens for evaluating mechanical properties: ultimate tensile strength, yield strength, elongation, and contraction (*a*) and impact strength (*b*)*1* – solid rolled ring, *2* – sheet material. Dimensions are given in mm**Рис. 2.** Эскизы образцов для оценки механических свойств: временного сопротивления, предела текучести, относительного удлинения и сужения (*a*) и ударной вязкости (*b*)*1* – цельнокатаное кольцо, *2* – листовой материал. Размеры указаны в мм

to determine the tensile strength ( $\sigma_u$ ), yield strength ( $\sigma_{0.2}$ ), relative elongation ( $\delta$ ), and contraction ( $\psi$ ) as per State Standard GOST 6996-66 (type XIII, Fig. 2, *a*). Impact strength (*KCU*) was also evaluated in accordance with the standard (type IX, Fig. 2, *b*). Notably, the weld was situated at the center of each specimen.

Grain size was determined by measuring chord lengths in accordance with State Standard GOST 5639-82 using specialized software. Microstructure analysis was conducted using an optical microscope equipped with a solid surface microstructure analyzer at various magnifications. Microhardness measurements were performed under a 50 g load.

## Results and discussion

Prior to welding, the tensile strength of specimens from a solid rolled ring was determined to be  $\sigma_u = 930\text{ MPa}$ , while the sheet material exhibited a tensile strength of  $\sigma_u = 990\text{ MPa}$ . Due to the higher strength of the latter, which exceeded  $980\text{ MPa}$ , a quenching process was carried out before welding. Fig. 3 provides insight into the microstructure of a 13 mm thick sheet of the heat-resistant dispersion-hardening nickel alloy EP718 in its as-delivered state and after quenching at  $1100\text{ }^{\circ}\text{C}$ , with a 2-hour hold, followed by air cooling. The microstructure of the base material post-heat treatment

consists of austenite grains, carbides, and the  $\gamma'$  phase (Fig. 3, *a*). In both scenarios, the average grain size falls within the range of 3–6, in accordance with State Standard GOST 5639-82.

A series of simulator specimens was produced under different welding modes (Table 3). These modes involved variations in the speed of beam movement along the specimen's surface (*v*) and the beam current (*i*). The selection of these parameters was guided by the specifications and aimed to ensure full penetration of specimens with a thickness of 13 mm. The results of the mechanical tests are provided in Table 4.

In modes 1 and 2, failure during the determination of time resistance occurred along the weld, whereas in other tested specimens, failure took place in the HAZ at a distance of 0.5–3.0 mm from the weld.

Table 3. Welding modes

Таблица 3. Режимы сварки образцов

Mode No.	<i>v</i> , m/s	<i>i</i> , mA
1	0.006	120
2	0.006	75
3	0.006	65
4	0.0042	85



Table 4. Mechanical properties of welded joints

Таблица 4. Механические свойства сварных соединений

Mode No.	$\sigma_u$ , MPa	$\sigma_y$ , MPa	$\delta$ , %	$\psi$ , %	KCU, MJ/m <sup>2</sup>	Microhardness, kg/mm <sup>2</sup>
1	1160	760	17.2	15.5	0.42	308
2	1170	770	17.8	19.0	0.41	329
3	1270	920	21.0	21.7	0.93	348
4	1160	810	15.4	15.5	0.81	335
Specifications TU 14-1-1059-2004	≥1080	≥790	≥13	—	≥0.35	293–363
Specifications TU 14-1-3905-85	≥1080	≥790	≥12	≥14	≥0.35	285–415

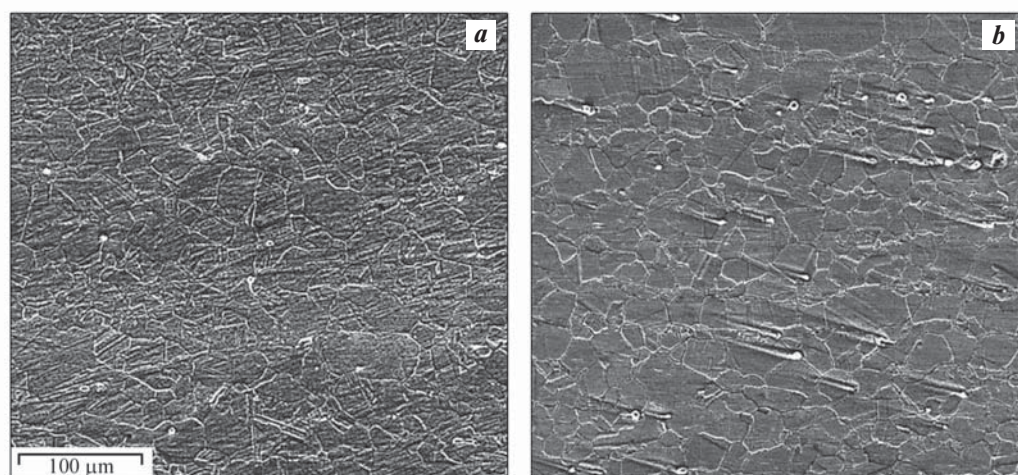
Fig. 3. Microstructure ( $\times 400$ ) of EP 718 sheet alloy in as-delivered condition (a) and after heat treatment (quenching) (b)

Рис. 3. Микроструктура ( $\times 400$ ) листового сплава ЭП718 в состоянии поставки (a) и после термической обработки (заковки) (b)

In the studied welds obtained in modes 1–3, complete penetration was achieved (Fig. 4, a–c). However, in the welded joint of mode 4, a minor lack of penetration of 0.1 mm was observed, with a single pore of 1.0 mm in diameter detected in one section (indicated by arrows in Fig. 4, d). According to specifications (RTM 1.4.1703-87), for this material thickness, pores of up to 1.5 mm are permissible. No cracks or other welding defects were identified during the micro-examination of sections of the welded specimens.

The macrostructure of the welds exhibits characteristic zones associated with EBW, including the “mushroom” zone (dimensions  $h$  and  $b$ ), the knife zone ( $e$  and  $d$ ), and the weld root in the locking part of the joints (Fig. 4, c and d) and in the lining area (Fig. 4, a). The overall dimensions of the weld for each mode are presented in Table 5.

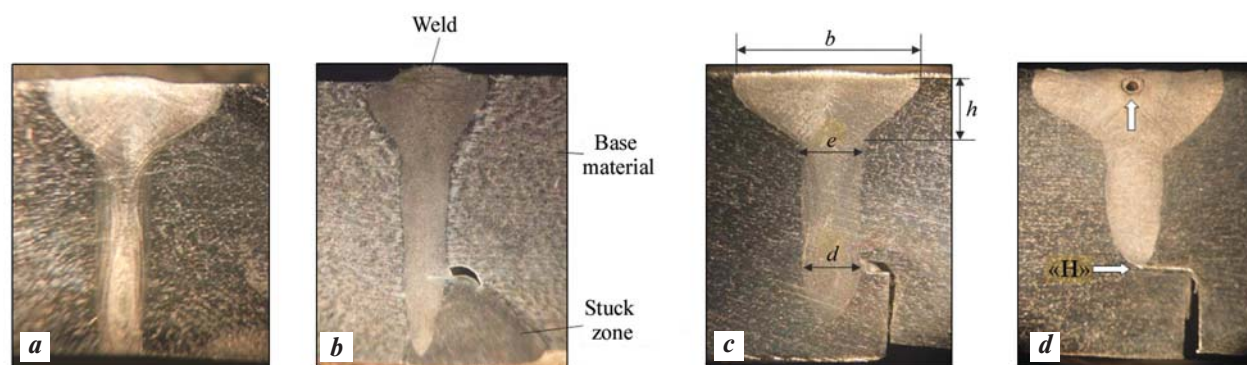
The highest weld penetration was achieved at a welding current of 120 mA, while specimens in modes 1–3 achieved the required depth. In welding mode 4, Table 5 shows the widest bath width ( $b$ ) in the “mushroom” zone and the shallowest depth of penetration. Mode 1 exhibits the highest “mushroom” zone height ( $h$ ), which can lead to additional heating, potential grain boundary thickening, and microcrack formation. Modes 2 and 3 yield the same “mushroom” zone height, but mode 2 has a narrower width. However, mode 3, with a beam current of 65 mA, provides the shallowest required weld depth, reducing heat input into the metal, minimizing grain boundary thickening in the heat-affected zone, and lowering the risk of crack formation.

In specimens welded in modes 1, 2, and 4, thickened grain boundaries are observed in HAZ before heat treat-

Table 5. Dimensions of the welds

Таблица 5. Габаритные размеры сварного шва

Mode No.	$h$ , mm	$b$ , mm	$e$ , mm	$d$ , mm	Penetration depth, mm
1	4.0–4.8	8.3–9.0	2.7–3.0	2.1–2.3	13.0
2	3.5–3.7	6.4–7.6	2.6–2.7	2.0–2.3	12.6
3	3.5–3.7	8.0–8.1	2.6–3.0	2.5	11.4
4	4.0–4.6	8.9–9.5	2.7–3.0	0.2–0.8	9.8
Specifications RTM 1.4.1703-87	Optional		1.5–3.5	1.5–2.5	$\geq 10.5$

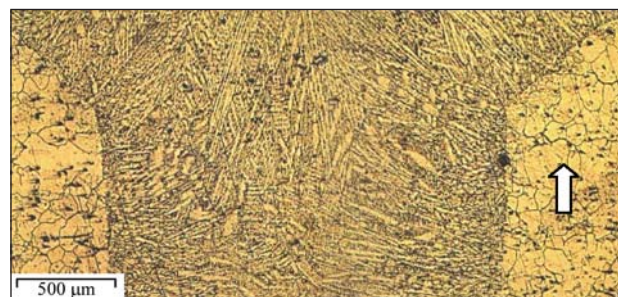
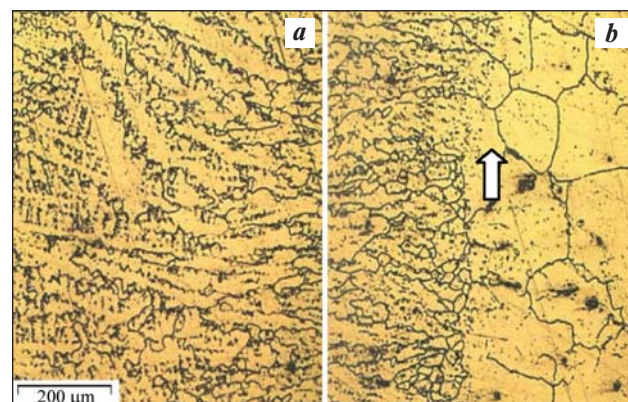
Fig. 4. Macrostructure of welds ( $\times 6$ ) obtained in modes 1 (a), 2 (b), 3 (c), and 4 (d)Рис. 4. Макроструктура сварных швов ( $\times 6$ ), полученных в режимах 1 (a), 2 (b), 3 (c) и 4 (d)

ment, likely due to carbide phase precipitation during welding and subsequent cooling, accelerating diffusion along grain boundaries, which can lead to crack formation [25]. After heat treatment, the number of thickened grain boundaries increases.

The microstructure of the base material of all specimens is heterogeneous, with varying grain sizes, mainly corresponding to 3–4 points on the GOST 5639-82 scale. In some areas and the heat-affected zone, individual grains with 2–3 points are observed. The micro-

structure of the specimen from welding mode 3, shown in Fig. 8, displays no defects in the form of cracks or lack of penetration.

In mode 3, optimal geometric weld parameters are achieved with no defects. This mode uses the lowest

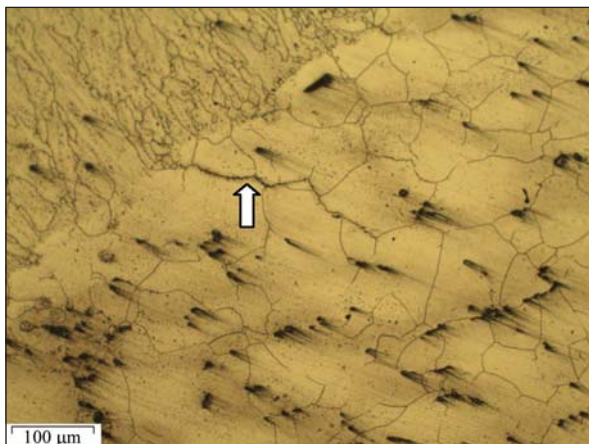
Fig. 5. Microstructure of the weld ( $\times 50$ ) obtained in welding mode 1Рис. 5. Микроструктура сварного шва ( $\times 50$ ), полученного в режиме сварки 1Fig. 6. Microstructure of the specimen ( $\times 200$ ) obtained in welding mode 2

a – weld, б – heat-affected zone

Рис. 6. Микроструктура образца ( $\times 200$ ), полученного в режиме сварки 2

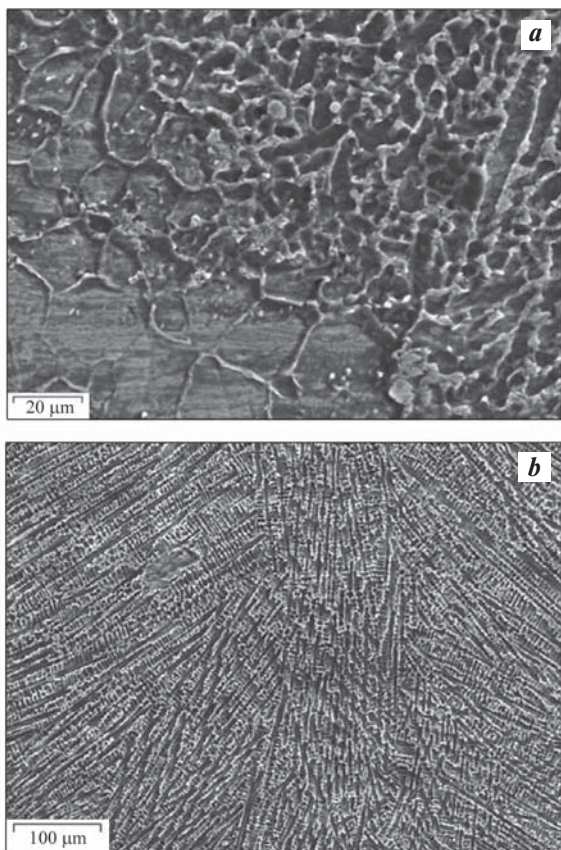
a – сварной шов, б – околосварная зона





**Fig. 7.** Microstructure ( $\times 100$ ) of the specimen in the heat-affected zone with thickened grain boundaries, obtained in welding mode 4

**Рис. 7.** Микроструктура ( $\times 100$ ) образца в околошовной зоне с утолщенными границами зерен, полученного в режиме сварки 4



**Fig. 8.** Microstructure of the base metal sample in the heat-affected zone (a) and the weld (b) during welding mode 3  
a –  $600\times$  magnification, b –  $250\times$  magnification

**Рис. 8.** Микроструктура образца основного металла в районе околошовной зоны (a) и сварного шва (b) при сварке в режиме 3  
a – увеличение  $600\times$ , b –  $250\times$

beam current, which minimizes base material heating while ensuring complete penetration of the edges. The resulting weld exhibits a columnar structure of a cast alloy, with thread-like crystals closer to the center, enhancing the material's heat resistance in the permanent joint.

The intense heat transfer during EBW conditions results in a reduction of the HAZ size and the formation of a fine dendritic structure along the weld's edge (refer to Fig. 8). The base metal in the HAZ maintains a uniform coarse-grained structure, incorporating carbonitrides, carbides, and a finely dispersed intermetallic  $\gamma'$  phase. In the microstructure of the HAZ, larger grains are observed compared to the base material.

## Conclusions

When studying welds in simulator specimens made of the alloy KhN45MVTYUBR-ID (EP718) with a thickness of 13 mm, the following observations were made:

1. The minimum tensile strength of 1160 MPa is recorded in specimens welded at  $v = 0.0042$  m/s and a beam current ( $i$ ) of 85 mA (mode 4). Conversely, specimens welded at  $v = 0.006$  m/s and  $i = 65$  mA (mode 3) demonstrated a maximum tensile strength of 1270 MPa.

2. Specimens welded at  $v = 0.006$  m/s, with  $i = 120$  and 75 mA (modes 1 and 2), exhibited fractures along the weld during tensile testing. In contrast, when using welding modes 3 and 4, destruction was observed along the HAZ at a distance of 0.5–3.0 mm from the weld.

3. Micro-examination of sections from specimens welded in modes 1, 2, and 4 revealed expanded grain boundaries in the HAZ.

4. Based on the obtained data, it was determined that welding mode 3, with parameters  $v = 0.006$  m/s and  $i = 65$  mA, is optimal. This mode exhibited no thickened grain boundaries and achieved an ultimate tensile strength of 1270 MPa.

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